

CHAPTER 3

STRESS IN A SOLID

AND

EQUILIBRIUM

THE PREVIOUS CHAPTER DEALS WITH THE KINEMATICS (GEOMETRY) OF DEFORMATION WITHOUT ADDRESSING THE AGENTS BEING RESPONSIBLE FOR IT. THE PRESENT CHAPTER IS DEVOTED TO INVESTIGATING WHAT CAUSES THE MOTION. INDEED, THIS STUDY PROVIDES A QUANTITATIVE METHOD TO DESCRIBE DEFORMABLE BODIES INTERACTION WITH THE HELP OF THE CONCEPTS OF FORCE, TRACTION VECTOR AND STRESS TENSOR.

AS OFTEN THE MAJOR CONTRIBUTOR TO FAILURE IS THE MAXIMUM STRESS, ELASTICITY THEORY PRIMARILY APPLICATION CONSISTS ON OBTAINING THE STRESS DISTRIBUTION ACROSS THE BODY. DUE TO THE CHARACTERISTICS OF MANY APPLICATIONS (THAT IS TRUE ALSO THAT IT IS MOTIVATED BY DIDACTICAL ISSUES), THE BODY IS CONSIDERED AT EQUILIBRIUM, WHICH IMPLIES THAT, CONSIDERING AN INFINITESIMAL ELEMENT AROUND ANY POINT, THE RESULTANTS MOMENT AND FORCE MUST BE ZERO.

IT IS WORTH MENTIONING THAT THE DEVELOPMENTS WITHIN THIS CHAPTER DO NOT REQUIRE THAT THE MATERIAL BE ELASTIC, AND THUS, IN PRINCIPLE, THE RESULTS APPLY TO A BROADER CLASS OF PROBLEMS.

3.1) Forces and Stress

DURING A MOTION MECHANICAL INTERACTIONS BETWEEN PARTS OF A BODY OR BETWEEN A BODY AND ITS ENVIRONMENT ARE

EXPRESSED THROUGH FORCES. IN GENERAL TERMS, WE RECOGNIZE THREE DIFFERENT TYPES OF FORCES, NAMELY,

(i) CONTACT FORCES BETWEEN "SEPARATE" PARTS OF A BODY (INTERNAL FORCES);

(ii) CONTACT FORCES EXERTED ON THE BOUNDARY OF A BODY BY EXTERNAL AGENTS; (SURFACE FORCES)

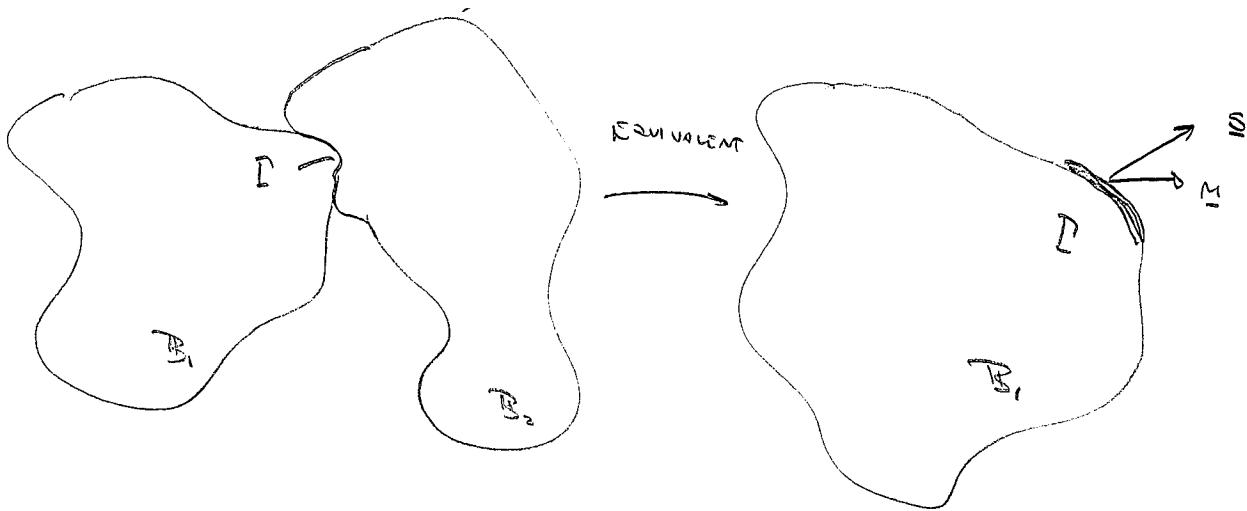
(iii) FORCES EXERTED ON THE INTERIOR POINTS OF A BODY BY THE ENVIRONMENT. (BODY (VOLUME) FORCES; e.g. GRAVITY)

} CAUCHY'S HYPOTHESIS:
(CONTACT FORCES)

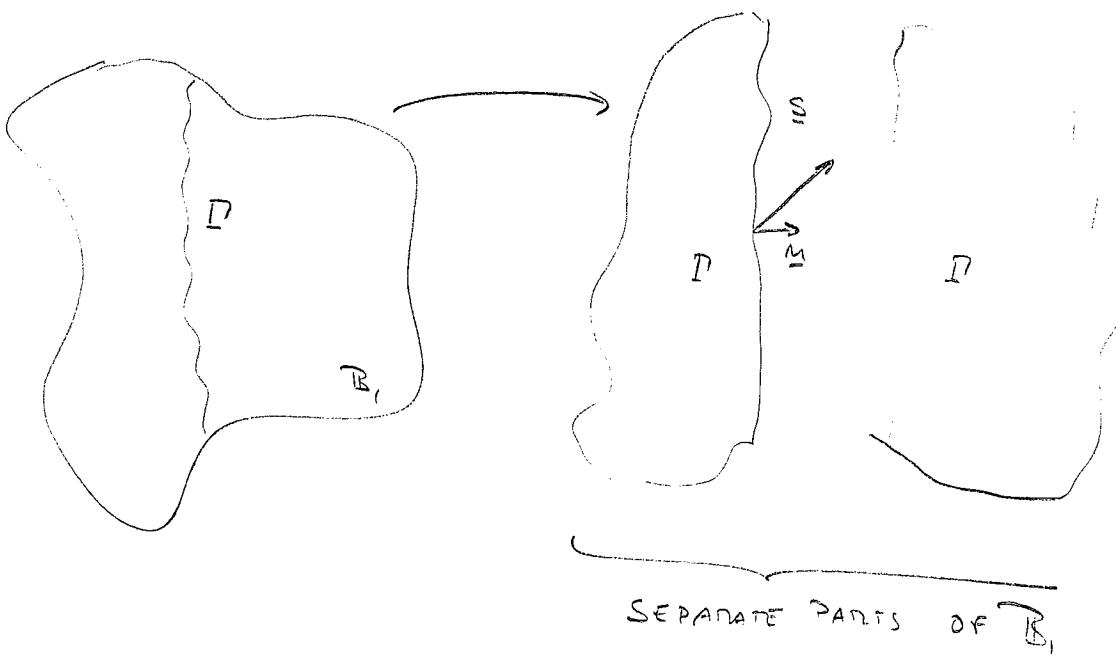
$\underline{s}(\underline{m}, \underline{x}, t)$ IS THE FORCE PER UNIT AREA,
EXERTED ACROSS \underline{l} (A ORIENTED SURFACE
WITH NORMAL GIVEN BY \underline{m}) UPON THE
MATERIAL ON THE NEGATIVE SIDE OF \underline{l}

Cauchy's Hypothesis Illustration

EXTERNAL Forces



INTERNAL Forces



Therefore, the resultant force and moment on a part P of B (note that P may coincide with B) is given by

$$\underline{F}(P) = \int_{\partial P} S d\underline{l} + \int_P b dV$$

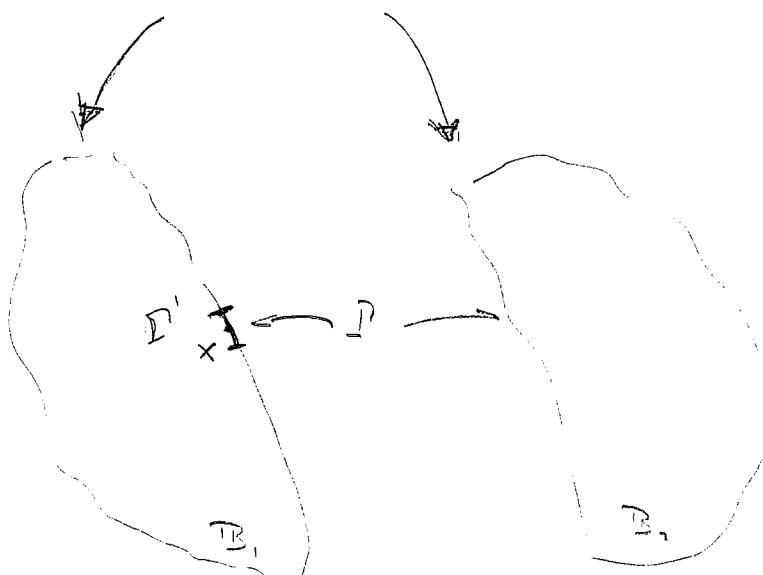
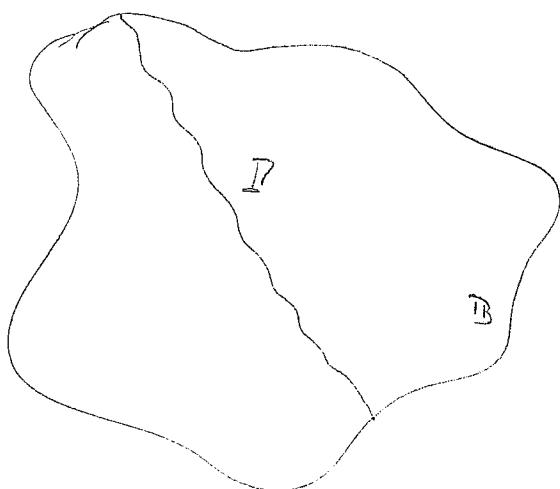
$$\underline{m}(P) = \int_{\partial P} (\underline{r} \times \underline{S}) d\underline{l} + \int_P (\underline{r} \times \underline{b}) dV$$

where b denotes the body force per unit volume, and \underline{r} is the position vector with respect to the origin of coordinates (remember \times denotes the vector product)

HEADING BACK TO INTERNAL FORCES, IT IS IMPORTANT TO INVESTIGATE

DEEPER ITS NATURE AND, FURTHER, QUANTIFY THEIR DISTRIBUTION

WITHIN THE SOLID BODY. SO, LET US AGAIN DIVIDE BODY B INTO TWO PARTS



(IT SEEMS SOMEHOW ARBITRARILY, AND IT IS ALTHOUGH OFTEN
GUIDED BY THE CIRCUMSTANCES OF THE APPLICATIONS)

LET US NOW FIX THE ATTENTION TO THE NEIGHBOR REGION OF x ON Γ (DENOTED AS Γ'). THE CONTACT RESULTANT FORCE ACTING ON Γ' IS

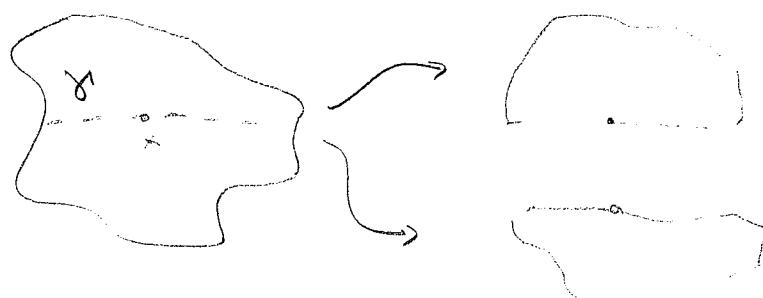
$$\underline{F}_{\Gamma'} = \int_{\Gamma'} s \, d\underline{l}$$

PASSING THROUGH A LIMITING PROCESS

$$\underline{F}(x) = \lim_{\Gamma' \rightarrow 0} \int_{\Gamma'} s \, d\underline{l}$$

OBTAINING, IN THAT WAY, THE FIGURE OF A PUNCTUAL FORCE.

Now, REMEMBER THE ABOVE RESULT WAS OBTAINED LAYING ON "AN ARBITRARY CHOICE" FOR Γ . Thus



NOTE THAT, AS s DEPENDS ON x AND WE WILL ARRIVE TO A DIFFERENT FORCE ACTING IN x .

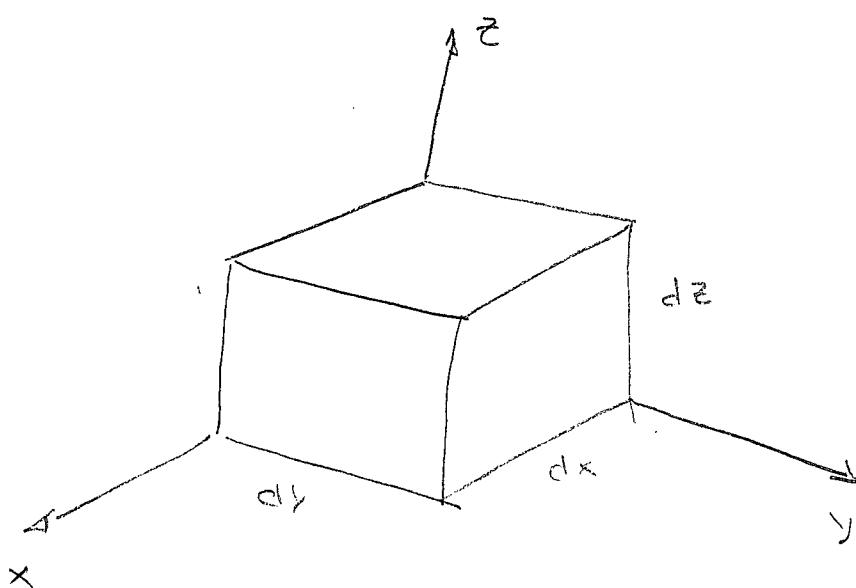
Therefore, the nature of internal forces is complex and can't be represented by a vector field.

Indeed, we introduce the vector

\underline{t} which is a function of the material point x and of the normal vector \underline{n} (which, in turn, defines the section of the body). Note that, from the Newton's action-reaction law

$$\underline{t}(x, \underline{n}) = -\underline{t}(x, -\underline{n})$$

Let us, once again, investigate the nature of a deformable solid by looking at an infinitesimal neighborhood of a point x taking a cubic form



THEN WE CAN COMPUTE \underline{t} THE VECTOR FOR ANY OF THE 6 FACES OF THE CUBE, LABELLING EACH ONE BY ITS NORMAL (e.g.: FACE X IS THE ONE WITH $\underline{m} = \underline{e}_x$; AND FACE -X CORRESPONDS TO $\underline{m} = -\underline{e}_x$):

$$\underline{t}(x, \underline{e}_x) = t_x^{e_x} \underline{e}_x + t_y^{e_x} \underline{e}_y + t_z^{e_x} \underline{e}_z$$

$$\underline{t}(x, \underline{e}_y) = t_x^{e_y} \underline{e}_x + t_y^{e_y} \underline{e}_y + t_z^{e_y} \underline{e}_z$$

$$\underline{t}(x, \underline{e}_z) = t_x^{e_z} \underline{e}_x + t_y^{e_z} \underline{e}_y + t_z^{e_z} \underline{e}_z$$

WHERE EACH SCALAR t_i^j IS THE i -TH COMPONENT OF THE \underline{t} VECTOR COMPUTED IN THE j -TH DIRECTION. WE WILL DISTINGUISH TWO SUB-SETS: NORMAL COMPONENTS (THAT WILL GIVE RISE TO THE LABEL NORMAL STRESS) AND SHEAR COMPONENTS (SHEAR STRESS). THE FORMER CONTAINS THE COMPONENTS IN THE \underline{m} DIRECTION. THE SHEAR COMPONENTS LIE ON THE SURFACE OF THE SECTION.

Considering now that $\underline{t} = \lim_{\overline{I} \rightarrow 0} \int_{\overline{I}} \underline{s} d\Gamma$ (so \underline{s} will be the traction vector, which reflects distribution of force within \overline{I})
the above expressions, we introduce the stress

tension \underline{T} such that its representation in the

coordinate frame is given by the matrix

$$[\underline{T}]_{ij} = \begin{bmatrix} s_x^{ex} & s_x^{ey} & s_x^{ez} \\ s_y^{ex} & s_y^{ey} & s_y^{ez} \\ s_z^{ex} & s_z^{ey} & s_z^{ez} \end{bmatrix}$$

(therefore it has units of Force per area)

It is straight forward to prove that for any section of the body defined by the normal \underline{m}

$$\underline{s}(x, \underline{m}) = \underline{T}(x) \underline{m}$$

(it is a good idea check the above result in one book of the reference list)

P.S: It will be shown later that the stress tensor is symmetrical and, by using the spectral theorem, it can be thus represented in the coordinate system formed by the principal directions (given by the eigenvectors) as

$$[\bar{\tau}] = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{bmatrix}$$

where σ_1, σ_2 and σ_3 are real numbers and the eigenvalues,
e.g.

$$\bar{\tau} \phi_i = \sigma_i \phi_i \quad (i=1,..3)$$

and, moreover, they are the principal stresses, which means that among σ_1, σ_2 and σ_3 we find the highest and lowest value for the stresses components. Note that in those directions the shear components are zero.

3.2

BALANCE LAWS AND EQUILIBRIUM

We restrict the present study to the motions observing equilibrium, which implies that in every and each arbitrary part of the body the applied loadings must satisfy the equations of equilibrium (viz.: the sum of all forces and the sum of all moments must vanish). If we, for instance, take the equations in page (3.4) and apply them to analyse the equilibrium of the whole body (Δ coincides with T_B) all we can find are forces acting on supports (reaction forces). The main objective of elasticity is to compute the stress and strain distribution within the body undergoing a deformable motion. Thus, we can partition it into appropriate subdomains and require that the part is ~~in~~ in equilibrium. So, let Δ a generic part inside T_B , than

$$\int_{\partial\Delta} s \, d\Gamma + \int_{\Delta} b \, dV = 0$$

AND

$$\int_{\partial P} \underline{S} \cdot d\underline{l} = \int_P \underline{T}_m \cdot d\underline{l}$$

BY THE DIVERGENCE THEOREM

$$\int_P \underline{T}_m \cdot d\underline{l} = \int_P \operatorname{div} \underline{T} dV$$

$$\text{So } \int_P (\operatorname{div} \underline{T} + \underline{b}) \cdot dV = 0$$

But the above result is valid for $\forall P \subset \mathbb{R}^3$, so,

once again, using the localization theorem:

$$\operatorname{div} \underline{T} + \underline{b} = 0 \quad \forall x \in \mathbb{R}$$

which is the local equilibrium equation (indeed

it is the balance of linear momentum disregarding inertial forces).

REPHERASING THE STRESS TENSOR IN COORDINATES, WE ADOPT
THE FOLLOWING CONVENTION

$$[\bar{\tau}] = \begin{bmatrix} \tau_{xx} & \tau_{xy} & \tau_{xz} \\ \tau_{yx} & \tau_{yy} & \tau_{yz} \\ \tau_{zx} & \tau_{zy} & \tau_{zz} \end{bmatrix}$$

WHERE τ_{ij} IS THE STRESS COMPONENT ACTING ON THE
i-th SURFACE AND IN j-th DIRECTION.

USUALLY, WE USE τ_i INSTEAD OF τ_{ii} ($\tau_x \sim \tau_{xx}$;

$$\tau_y \sim \tau_{yy} \text{ AND } \tau_z \sim \tau_{zz})$$

VERY SOON, WE WILL SEE THAT $\tau_{ij} = \tau_{ji}$

THE ABOVE EQUATION CORRESPONDS TO THREE SCALAR EQUATIONS
IN A CARTESIAN COORDINATE FRAME

$$\frac{\partial \tau_x}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z} + b_x = 0$$

$$\frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{zy}}{\partial z} + b_y = 0$$

$$\frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \tau_{xx}}{\partial z} + b_z = 0$$

INVOLVING NINE UNKNOWNS.

NOW CONSIDERING THE ANGULAR MOMENTUM EQUATION (THE
BALANCE OF MOMENTS LEADING TO MUST VANISH IN
EQUILIBRIUM) WE HAVE

$$\int_P (\underline{r} \wedge \underline{s}) d\underline{r} + \int_P (\underline{r} \wedge \underline{b}) d\underline{r} = 0$$

$$\int_{\partial P} (\underline{n} \wedge \underline{s}) d\underline{l} = \int_{\partial P} (\underline{n} \wedge T \underline{m}) d\underline{l} =$$

$$= \int_{\partial P} (RT) \underline{m} d\underline{l} = \int_P \operatorname{div}(RT) dA$$

()

DIVERGENCE THEOREM!

WHERE T IS AN ANTI-SYMMETRIC SUCH THAT $T \underline{m} = \underline{n} \wedge \underline{m}$

WITH $[T] = \begin{bmatrix} 0 & -z & y \\ z & 0 & -x \\ -y & x & 0 \end{bmatrix}$ AND THEN -

$$[RT] = \begin{bmatrix} -z \tau_{ay} + y \tau_{az} & -z \tau_{yy} + y \tau_{yz} & -z \tau_{zy} + y \tau_{zz} \\ z \tau_{ax} - x \tau_{ay} & z \tau_{yx} - x \tau_{yz} & z \tau_{zx} - x \tau_{zy} \\ -y \tau_{ax} + x \tau_{ay} & -y \tau_{xy} + x \tau_{yy} & -y \tau_{xz} + x \tau_{zy} \end{bmatrix}$$

AND

$$\left[\begin{array}{c} -z \frac{\partial \tau_{xy}}{\partial x} + y \frac{\partial \tau_{xz}}{\partial x} - z \frac{\partial \tau_{yy}}{\partial y} + \tau_{yz} + y \frac{\partial \tau_{yz}}{\partial y} - \tau_{zy} - z \frac{\partial \tau_{zy}}{\partial z} + y \frac{\partial \tau_{zz}}{\partial z} \\ \\ z \frac{\partial \tau_{xz}}{\partial x} - \tau_{yz} - x \frac{\partial \tau_{xz}}{\partial x} + z \frac{\partial \tau_{yz}}{\partial y} - x \frac{\partial \tau_{yz}}{\partial y} + \tau_{zy} + z \frac{\partial \tau_{zy}}{\partial z} - x \frac{\partial \tau_{zz}}{\partial z} \\ \\ -y \frac{\partial \tau_{xx}}{\partial x} + \tau_{xy} + x \frac{\partial \tau_{xy}}{\partial x} - \tau_{yx} - y \frac{\partial \tau_{yx}}{\partial y} + x \frac{\partial \tau_{yx}}{\partial y} - y \frac{\partial \tau_{zz}}{\partial z} + x \frac{\partial \tau_{zz}}{\partial z} \end{array} \right]$$

WHICH CAN BE DECOMPOSED INTO

$$\text{div}(\underline{\tau} \cdot \underline{T}) = \left[\begin{array}{c} \tau_{yz} - \tau_{zy} \\ \tau_{xy} + \tau_{zy} \\ \tau_{xy} - \tau_{yz} \end{array} \right] + (\underline{n} \wedge \text{div} \underline{T})$$

(3.10)

PLUGGING THIS EXPRESSION IN THE ANGULAR MOMENTUM

BALANCE WE HAVE

$$\oint_{\Gamma} \underline{r} \times (\text{div } \underline{T} + \underline{b}) d\underline{l} + \int_{A} \underline{b} \cdot \underline{d\underline{r}} = 0$$

BUT NOTE THAT IN THE FIRST INTEGRAL WE HAVE

THE BALANCE OF LINEAR MOMENTUM, SO THE INTEGRAL
VANISHES AND

$$\int_{A} \underline{b} \cdot \underline{d\underline{r}} = 0 \quad \downarrow \quad \forall p \in \mathbb{R}$$

$$\int_{\Gamma} \underline{b} \cdot \underline{d\underline{l}} = 0 \rightarrow \begin{cases} T_{y_3} = F_{3y} \\ T_{z_3} = F_{3z} \\ F_{ay} = F_{y_2} \end{cases}$$

$$\boxed{\overline{T} = T^+}$$

Comment: The proof of T being symmetric should be carried out by using the result contained in the example of Pg. 147. How?

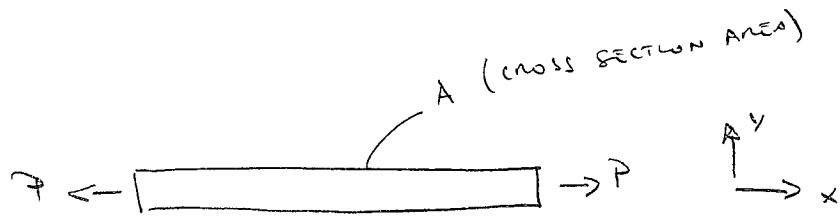
P.S: (i) Due to the adoption of the "LAGRANGIAN view" of the deformation the mass balance is automatically satisfied

(ii) Indeed, all the integrals associate to the balance laws must be performed over $\chi(p) \in \chi(\mathbb{B})$

(regions assigned in the deformed configuration). But, within the realm of small deformations, \mathcal{P} and $\chi(p)$ (as \mathbb{B} and $\chi(\mathbb{B})$) are close and, for certain purposes, can be idealized as the same region.

3.3) EXAMPLES

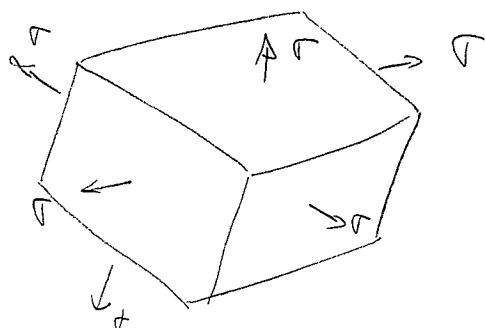
(1) UNIAXIAL STRESS



$$[T] = \begin{bmatrix} P/A & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

PRINCIPAL STRESSES: $\sigma_1 = P/A > \sigma_2 = \sigma_3 = 0$

(2) HYDROSTATIC STRESS

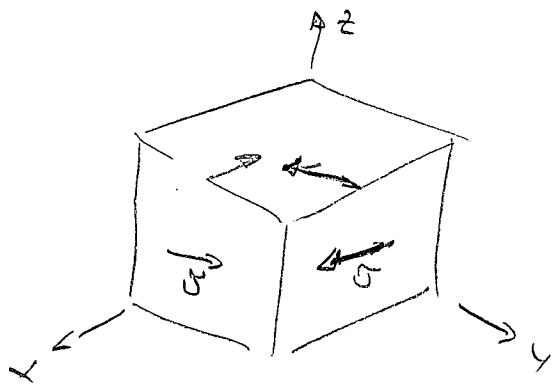


PRINCIPAL STRESSES:

$$\sigma_1 = \sigma_2 = \sigma_3 = P_r$$

$$[T] = \begin{bmatrix} P_r & 0 & 0 \\ 0 & P_r & 0 \\ 0 & 0 & P_r \end{bmatrix} = P_r I$$

(3) Pure Shear



$$[T] = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & \sigma & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

PRINCIPAL STRESSES:

$$[T] \Phi_i = \sigma_i \Phi_i$$

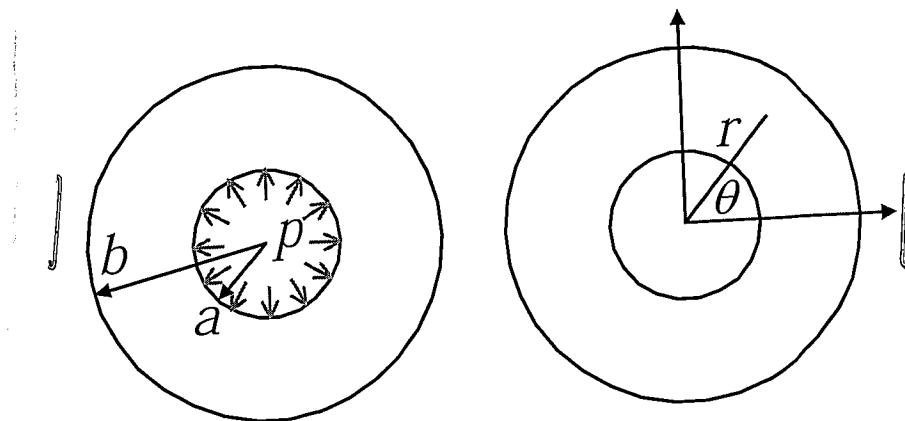
CHARACTERISTIC EQUATION:

$$\det [T - \sigma_i I] = 0$$

$$-\sigma_i(\sigma_i^2 - \gamma^2) = 0 \implies \sigma_i = \gamma ; \sigma_{II} = 0 ; \sigma_{III} = -\gamma$$

Therefore the principal stresses are localized in the XY plane pointing towards a 45° direction.

(4)



At the inner boundary ($r=a$) the traction vector is given by

$$\underline{\sigma} = \underline{T}(-\underline{e}_r) = P \underline{e}_r$$

↳ (why?)

And at $r=b$

$$\underline{\sigma} = \underline{T}(\underline{e}_r) = \underline{0}$$

Therefore,

$$\text{At } r=a \rightarrow \underline{\tau}_{rr} = \underline{e}_r \cdot \underline{T}_{er} = -P$$

$$\underline{\tau}_{r\theta} = \underline{e}_\theta \cdot \underline{T}_{er} = 0$$

$$\text{At } r=b \rightarrow \underline{\tau}_{rr} = 0 ; \underline{\tau}_{r\theta} = 0$$

P_{PRINCIPAL} D_{IRECTIONS} (EIGEN VECTORS)

I ($\Gamma_I = \gamma$)

$$\begin{bmatrix} -\gamma & \gamma & 0 \\ \gamma & -\gamma & 0 \\ 0 & 0 & -\gamma \end{bmatrix} \begin{bmatrix} \phi_1^x \\ \phi_1^y \\ \phi_1^z \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$

$$\rightarrow \phi_1^3 = 0 \quad ; \quad \phi_1^x = \phi_1^y$$

IF we take $\|\phi_i\|^2 = 1 \rightarrow \phi_1 = \begin{bmatrix} \pm 1/\sqrt{2} \\ \pm 1/\sqrt{2} \\ 0 \end{bmatrix}$

II ($\Gamma_{II} = 0$)

$$\phi_2^x = \phi_2^y = 0$$

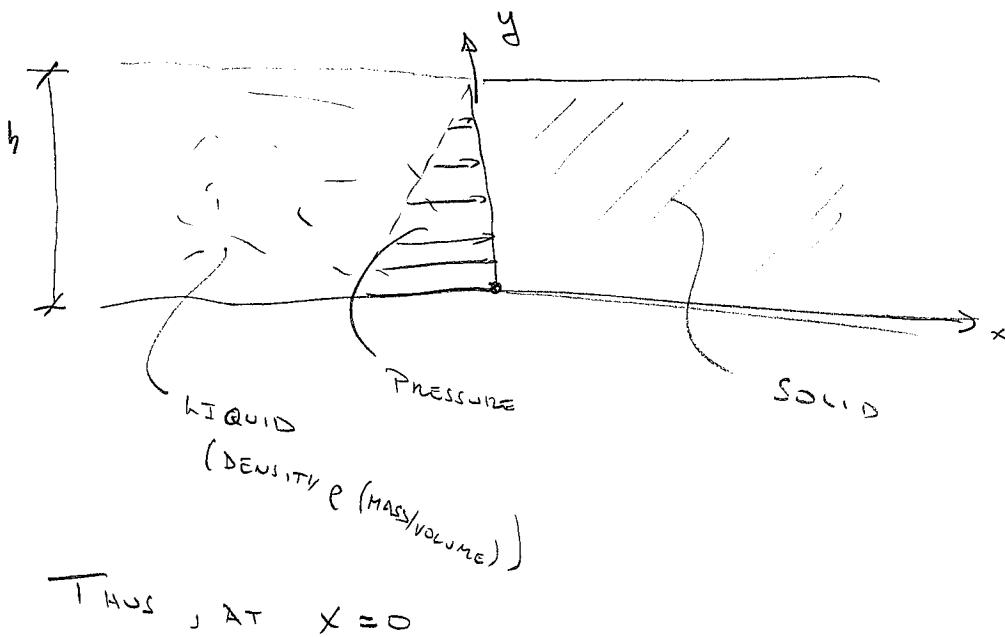
AND THERES, FROM THE NORMALIZATION, $\phi_2^3 = \pm 1$

III ($\Gamma_{III} = -\gamma$)

$$\phi_3^x = -\phi_3^y \quad \text{AND} \quad \phi_3^3 = \omega$$

$$\phi_3 = \pm \begin{bmatrix} \frac{1}{\sqrt{2}} \\ -\frac{1}{\sqrt{2}} \\ 0 \end{bmatrix}$$

(15)



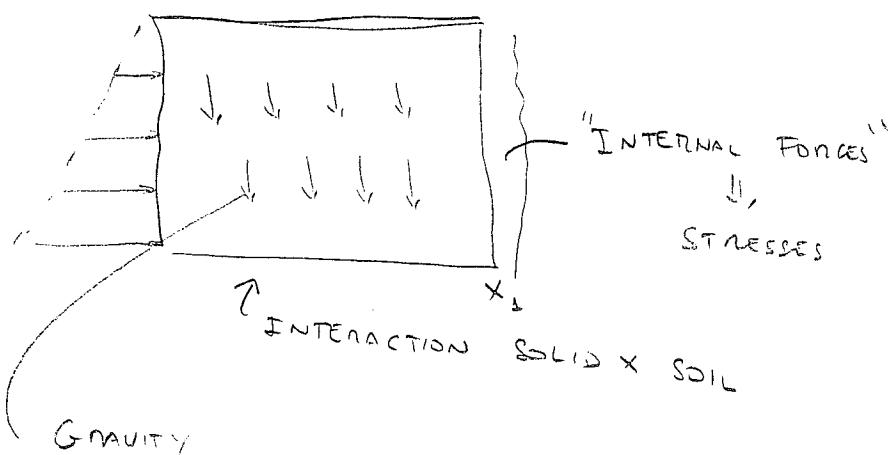
$$\underline{\sigma}(0, y) = \underline{T}(0, y) (-e_1) = \rho g (h - y) e_1$$

At $x=0 \rightarrow T_{2x} = -\rho g (h - y) \quad (\text{COMPRESSION})$

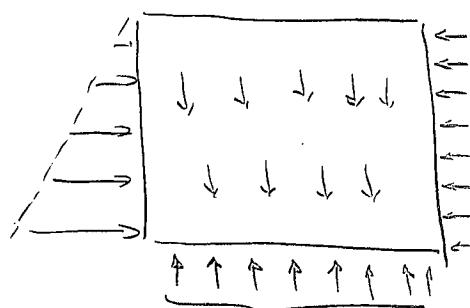
$$T_{2y} = 0$$

WHAT HAPPENS IN THE SECTION OF THE SOLID DEFINED

BY $x = x_s$



ASSUMING THAT THERE IS NO FRICTION BETWEEN SOLID AND SOIL AND UNIFORMITY ALONG Z DIRECTION (\Rightarrow 2D PROBLEM)



REACTION = FORCE LINEAR DISTRIBUTION R

* EQUILIBRIUM IN THE Y-DIRECTION

$$\text{WITH } b = -\rho_s g e_y$$

\hookrightarrow (SOLID DENSITY)

$$\rho_s g (x_1 h) = \int_A b \cdot e_y d\Gamma = \int_{\Gamma} S \cdot e_y d\Gamma = \int_0^{x_1} R \cdot e_y dx = R x_1$$

AND THEREFORE :
$$\boxed{T_{yy}(x_1, 0) = -\rho_s g h}$$

IF $x < x_1$

INDEED $T_{yy}(x_1, 0) = -\rho_s g h$ ALWAYS!

AND MORE:

$$\tau_{yy}(x, y) = -\rho_s g y$$

EQUILIBRIUM IN THE X-DIRECTION

$$\int_P S \cdot e_x \, dP = 0$$

AND THEN

$$-\int_0^h \tau_{xx} \Big|_{z=0} dy + \int_0^h \tau_{xx} dy = 0$$

$$\text{THEN } \tau_{xx}(x_1, y) = -\rho g (h-y)$$

AND AS x_1 IS ARBITRARY

$$\tau_{xx}(x, y) = -\rho g (h-y)$$